

# Generation of large-scale curved meshes for complex virtual geometries

**Eloi Ruiz-Gironés with Xevi Roca**

Tetrahedron VII Workshop



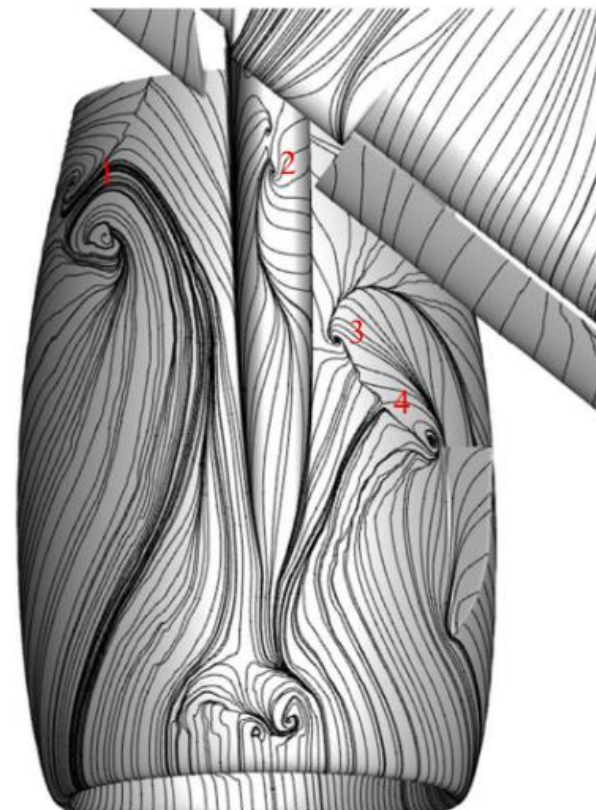
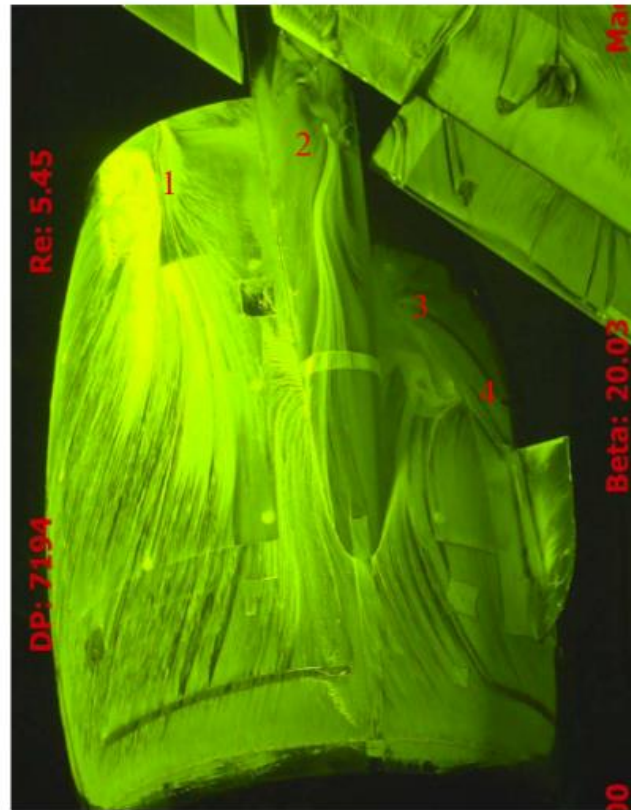
**European Research Council**  
Established by the European Commission

# Acknowledgements

- **Co-authors:**
  - J. Sarrate (UPC), A. Gargallo (BSC) & X. Roca (BSC)
- **Simulations on our curved meshes:**
  - 4<sup>th</sup> & 5<sup>th</sup> HLPW high-order group
  - Z.J. Wang, University of Kansas, USA
  - Oriol Lehmkuhl, BSC, Spain
- **Computing hours:**
  - BSC
  - PRACE program

# Motivation: Wind Tunnel vs Simulation

Wind tunnel data  
(4<sup>th</sup> HLPW)  
(NASA AIAA)



High-order simulation  
(ZJ Wang AIAA'23)

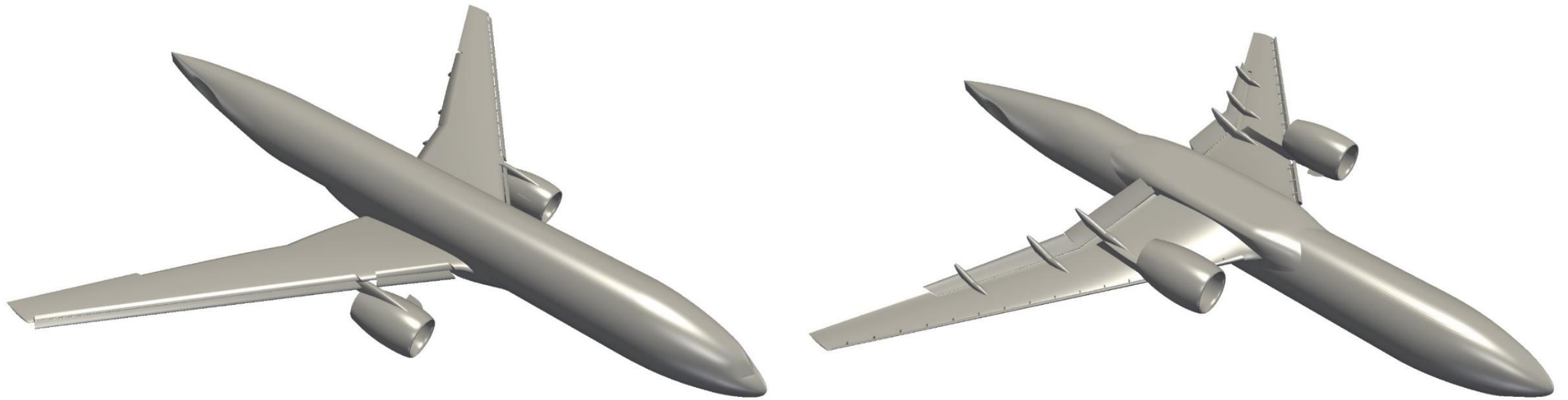
## Unstructured high-order methods

- **Geometric flexibility:** using unstructured meshes
- **More accuracy:** with same #DOFs, less dissipation and dispersion

## They require curved high-order meshes

- **Geometric error:** straight elements hamper simulation accuracy

# Motivation: Curved Meshes for Flow Simulation



## Large-scale curved meshes in complex virtual geometries

- **Approximate virtual CAD B-rep:** using curved elements
- **Mesh features:** small elements & size gradation, boundary layer...
- **Mesh quality:** facilitates solving the simulation problems

# Mesh Curving Methods

**Direct methods:** Create curved mesh from scratch

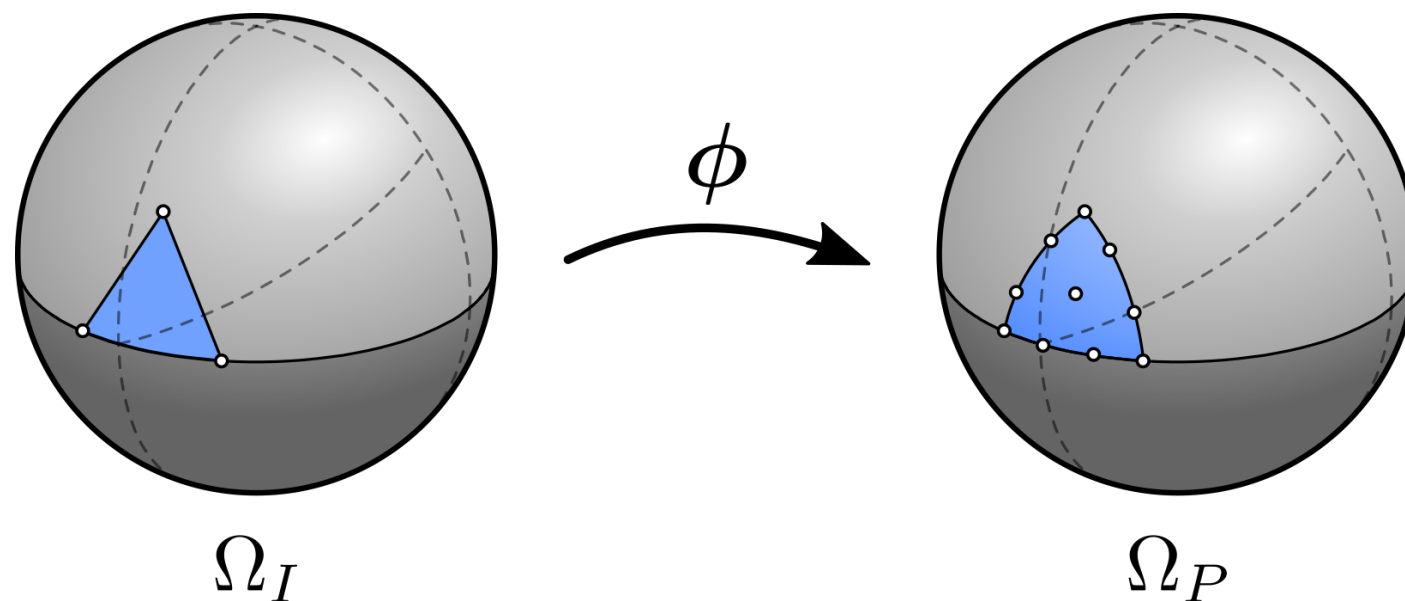
- Delaunay (Feng, Alliez, Busé, Delingette, Desbrun ToG'18)
- Advancing front (Mohammadi, Shontz IMR'21)

**Indirect methods:** Linear mesh generation + curving step

- **PDE-based**
  - Linear / non-linear elasticity  
(Persson, Peraire), (Xie, Poya, Sevilla, Hassan), (Turner, Moxey, Sherwin, Peiró)
  - Winslow (Fortunato, Persson), ...
- **Optimization-based**
  - Mesh distortion / quality (Tomov, Mittal, Kolev), (Karman),  
(Gargallo-Peiró, Ruiz-Gironés, Sarrate, Roca), (Feuillet, Loseille, Alauzet)...
  - Nodal displacement (Toulorge, Johnen, Lambrechts, Remacle), ...
  - Other quantities (Stees, Shontz), ...



- **Curving solution**
- **Complex geometry in parallel**
- **Large-scale distributed curving**
- **High-Lift Prediction Workshop**



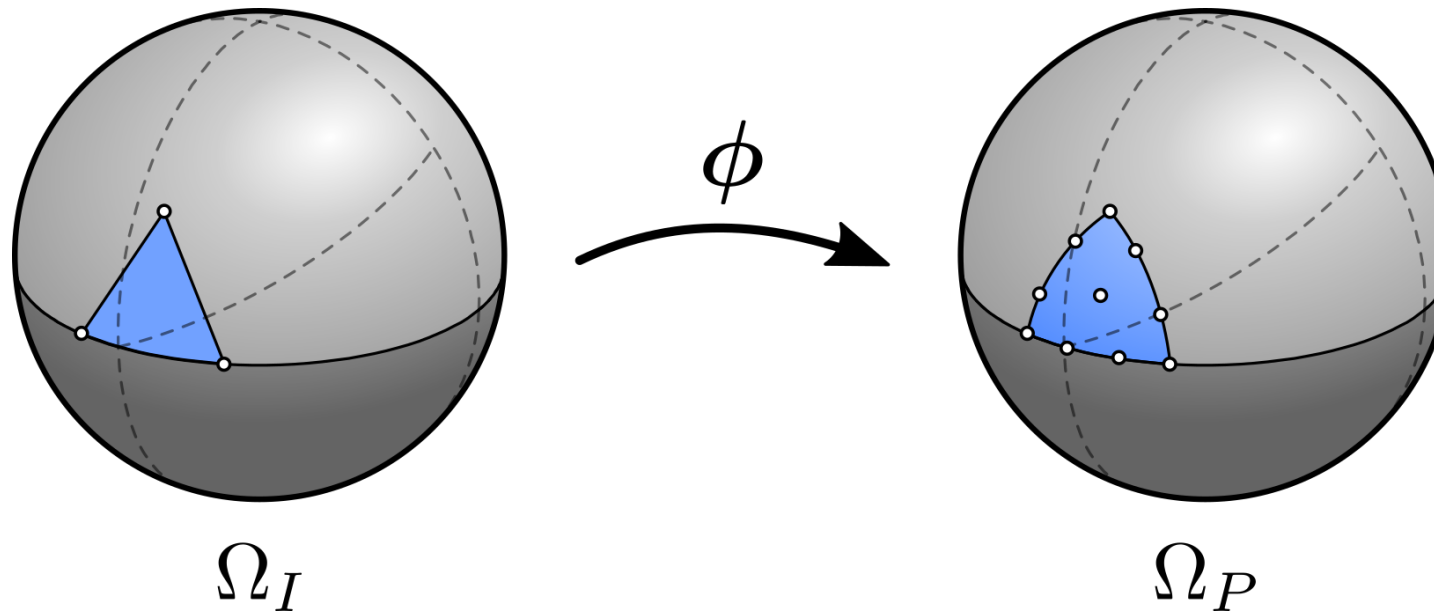
# Curving Solution

(Ruiz-Gironés, Gargallo, Sarrate, Roca IMR'17 & CAD'19)

(Ruiz-Gironés, Roca IMR'18, IMR'19 & CAD'22)

- **Constrained optimization problem**
- **Continuous penalty**

# Our Formulation : Constrained Optimization



$$M\phi = \frac{|\mathbf{D}\phi|^2}{n\sigma_0(\mathbf{D}\phi)^{2/n}}$$

(Knupp SIAM J. Sci. Comput.'01)  
 (Roca, Gargallo, Sarrate IMR'12)  
 (Gargallo, Roca, Peraire, Sarrate IJNME'15)

$$\min_{\phi} \|M\phi\|_{\Omega_I}^2$$

constrained to:

$$\mathbf{T}\phi = \mathbf{g}_D(\mathbf{T}\phi)$$

(Ruiz-Gironés, Roca, Sarrate CAD'16)  
 (Ruiz-Gironés, Gargallo, Sarrate, Roca IMR'17 & CAD'19)  
 (Ruiz-Gironés, Sarrate, Roca IMR'16)

(Ruiz-Gironés, Sarrate, Roca IMR'15 & JCP'21)

$$\mathbf{g}_D(\mathbf{T}\phi) = \sum_{i=1}^N \pi_{\Omega_i}(\mathbf{x}_i) N_i$$

(Ruiz-Gironés, Roca IMR'18, IMR'19, CAD'22, AIAA'22)



# Our Solution: Continuous Penalty Method

(Ruiz-Gironés, Roca IMR'18, IMR'19 & CAD'22)

**Penalty method:** Optimize several unconstrained problems

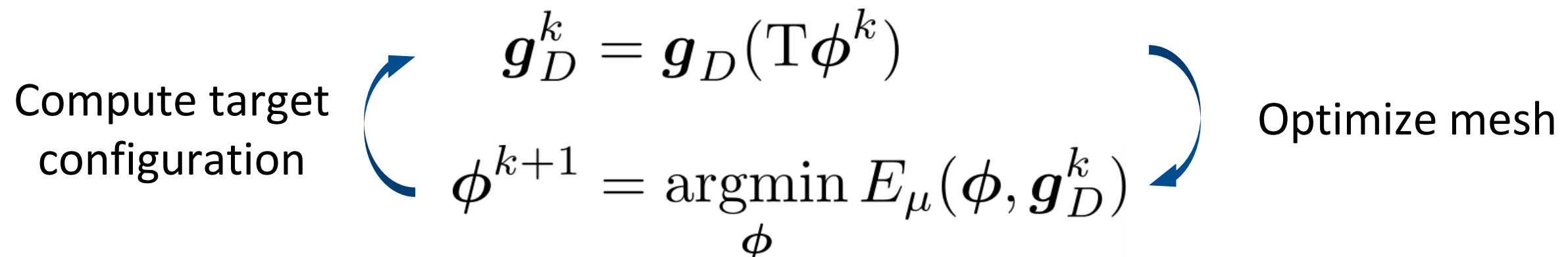
$$E_\mu(\phi) = ||M\phi||_{\Omega_I}^2 + \mu ||T\phi - \mathbf{g}_D(T\phi)||_{\partial\Omega_I}^2$$

$$E_\mu : \mathcal{H}^1(\Omega) \longrightarrow \mathbb{R}$$

$$\mathbf{g}_D : \mathcal{L}^2(\partial\Omega) \longrightarrow \mathcal{L}^2(\partial\Omega)$$

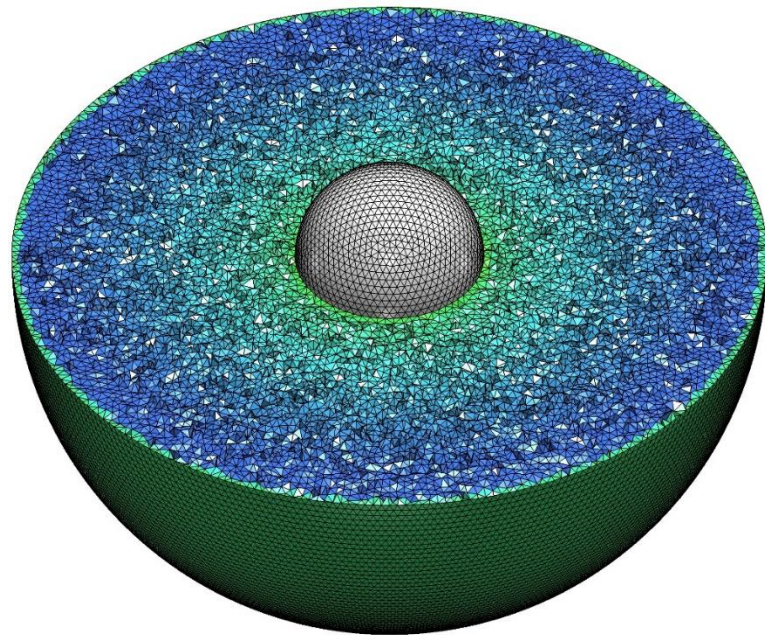
**Non-linear problem:** volume & boundary

**Fix-point iterative solver:** Newton + backtracking line-search

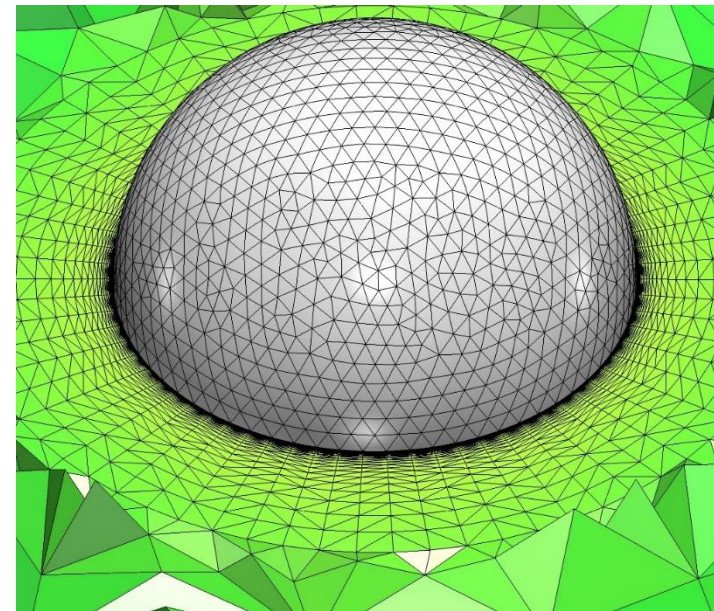




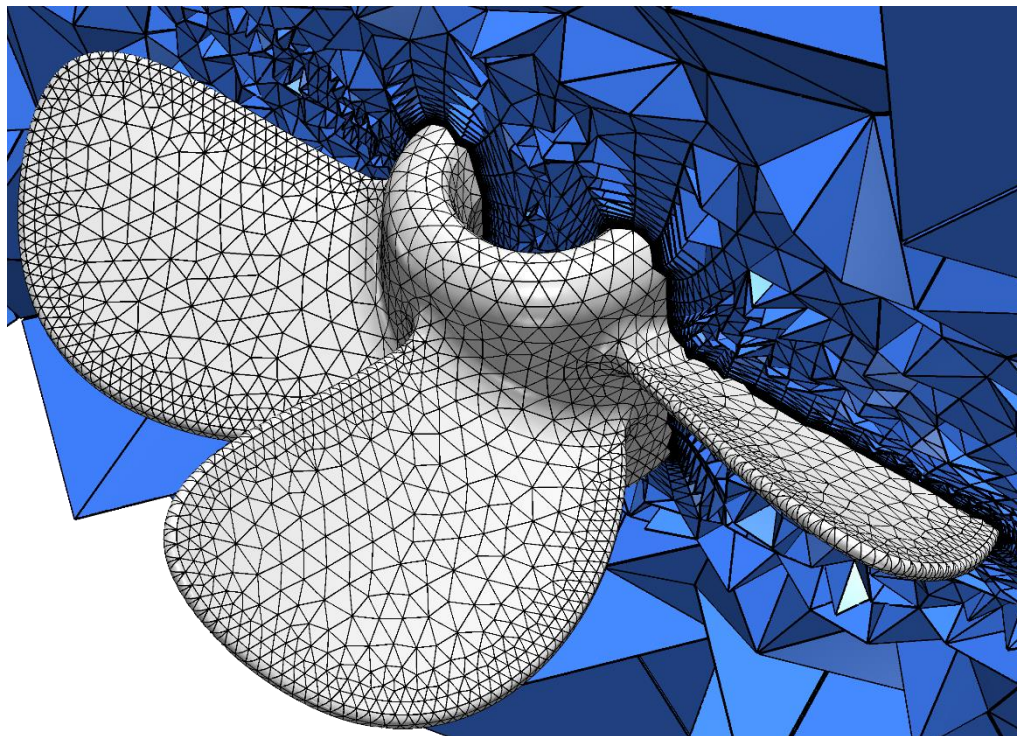
# Example: Sample meshes



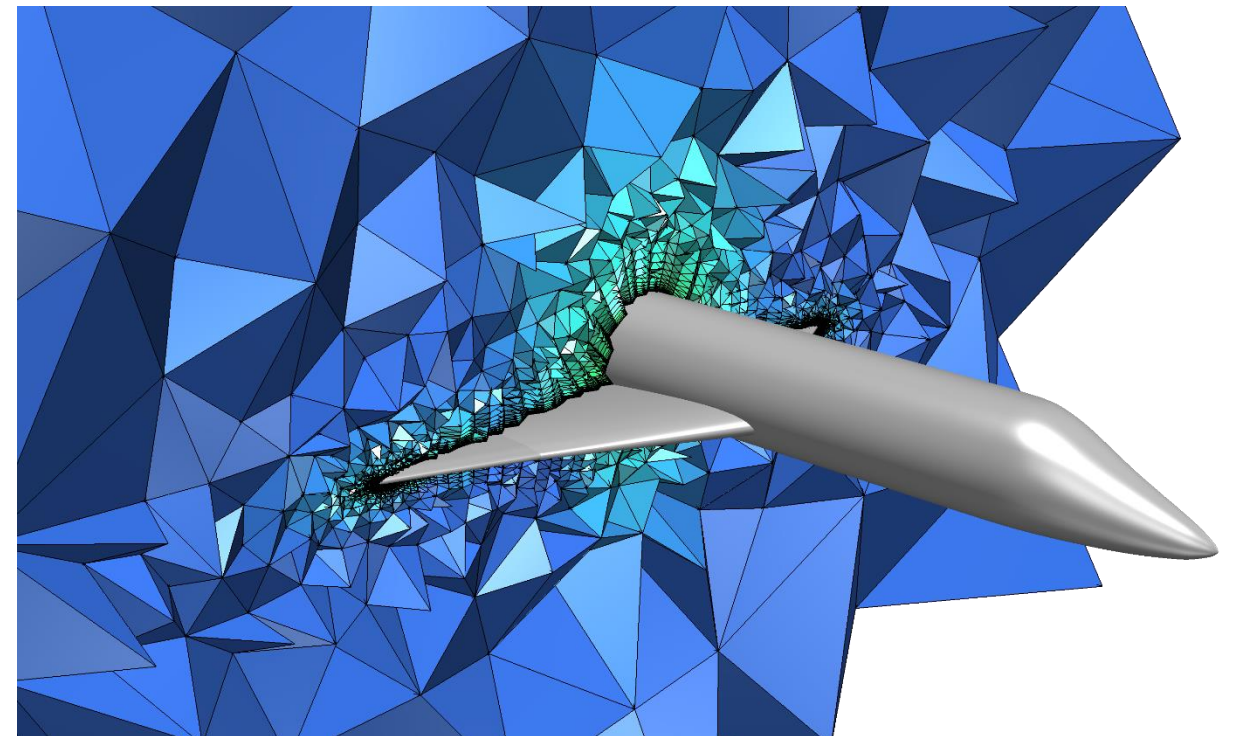
3.6M tets,  $p=4$



0.7M tets,  $p=4$   
stretching  $1:10^5$



1.6M tets,  $p=3$ , stretching 1:750

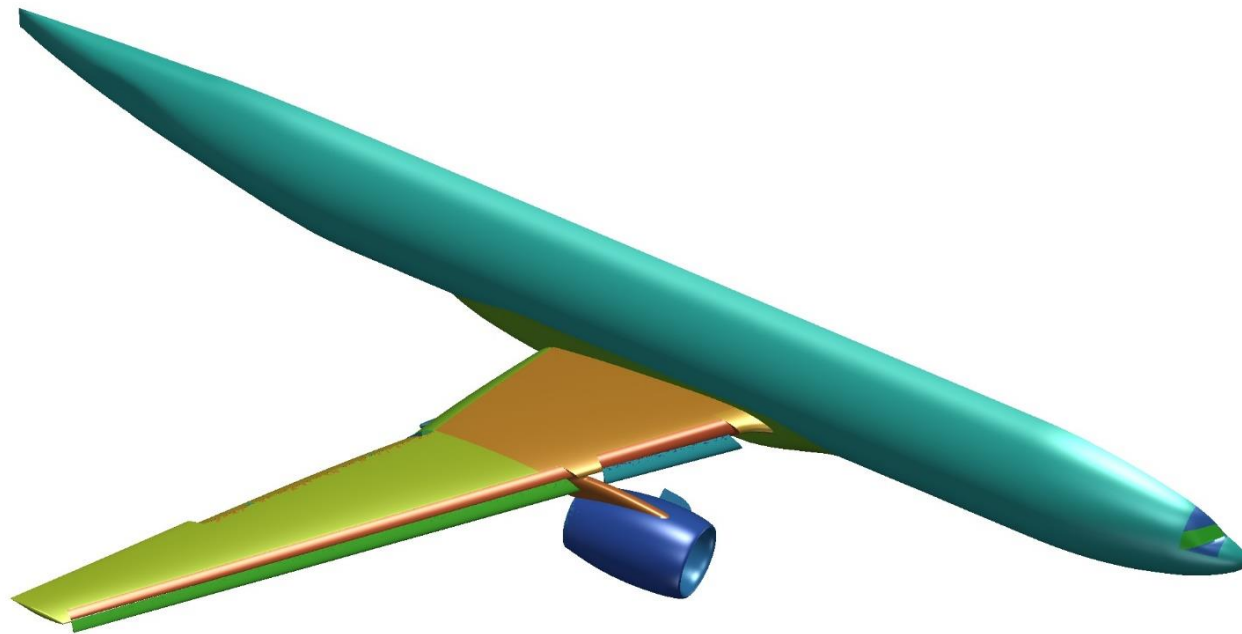


4M tets,  $p=4$ , stretching 1:400



# Features

- **Constrained optimization problem:**  
Minimize mesh distortion while approximating the virtual model
- **Mesh floats:**  
Mesh approximates the geometry
- **Mesh is always valid:**  
No need to introduce untangling
- **Virtual geometry aware:**  
Elements span several entities
- **Tight tolerances:**  
Fully converged meshes avoid element oscillations
- **Newton's method with backtracking line-search:**  
Ensures quadratic convergence near solution



# Complex geometry in parallel

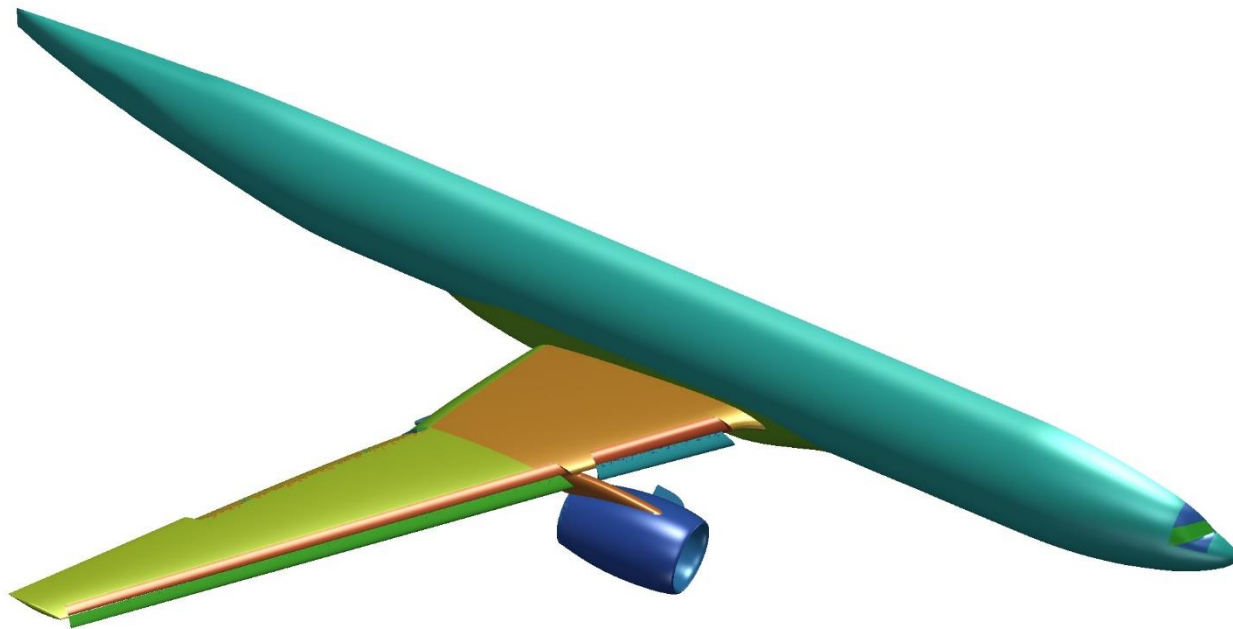
(Ruiz-Gironés, Roca IMR'18, IMR'19 & CAD'22)

(Ruiz-Gironés, Roca AIAA'22)

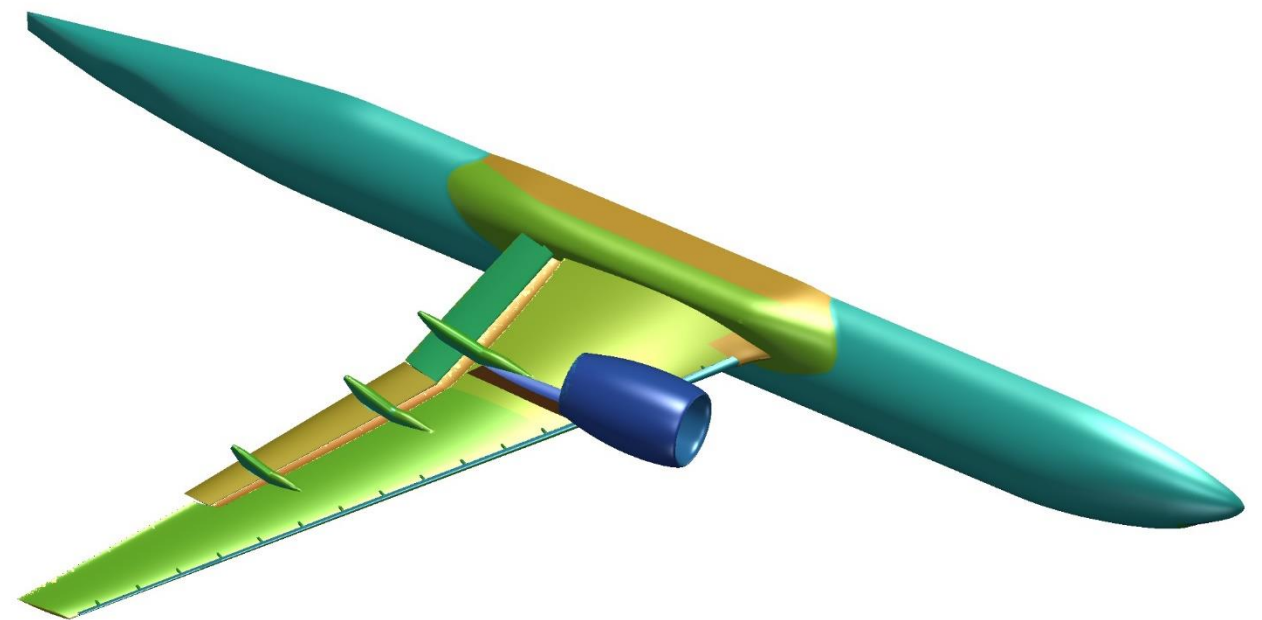
- **Virtual model**
- **Point projection**
- **Parallel distribution of input & output**

# Virtual Model

- **Virtually join surfaces:** Simulation intent
- **Decouple CAD & mesh topologies:** one group for fuselage, wings, ...
- **Surfaces:** From 415 original surfaces to 215 virtual surfaces



Virtual surfaces: top view



Virtual surfaces: bottom view

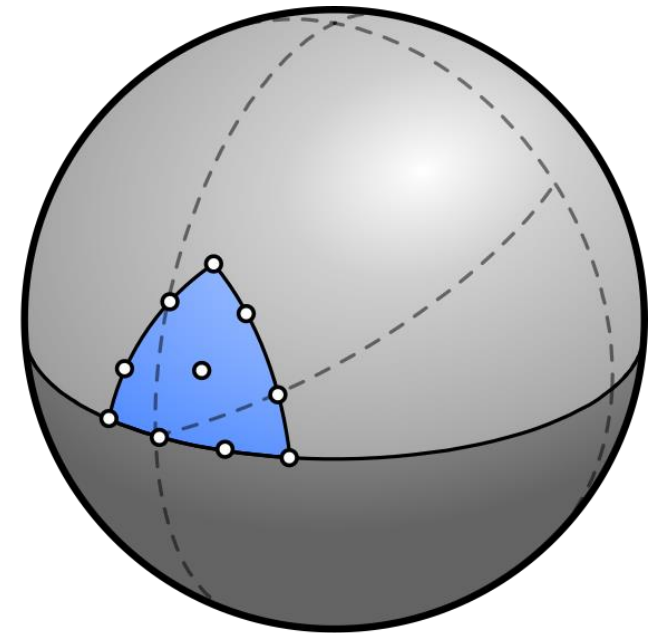
# Point Projection onto Virtual Models

- **Projection onto virtual surface:**

(Ruiz-Gironés, Roca IMR'18, IMR'19 & CAD'22)

- Loop over the geometric surfaces

$$\Pi_{\mathcal{S}}(\mathbf{x}) = \arg \min_{\mathbf{y} \in \mathcal{S}} \|\mathbf{x} - \mathbf{y}\|$$



- **Projection onto virtual curve:** Dealing with surface gaps

(Ruiz-Gironés, Roca AIAA'22)

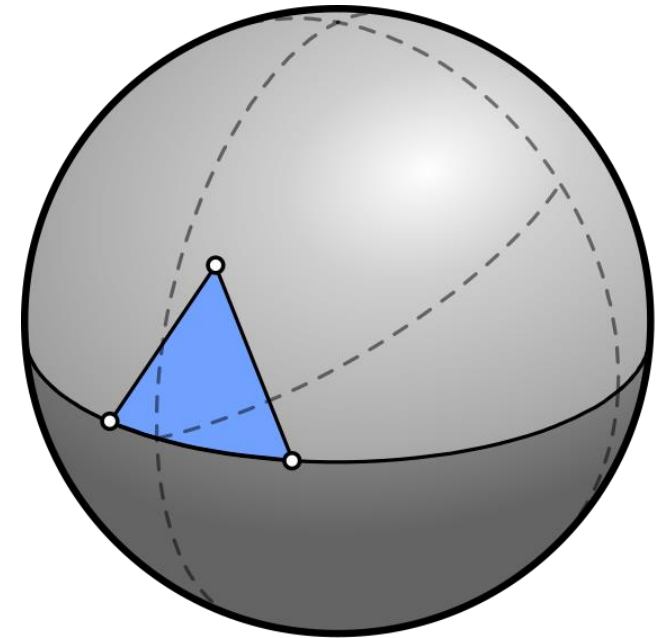
- Project the node in-between the surface gap

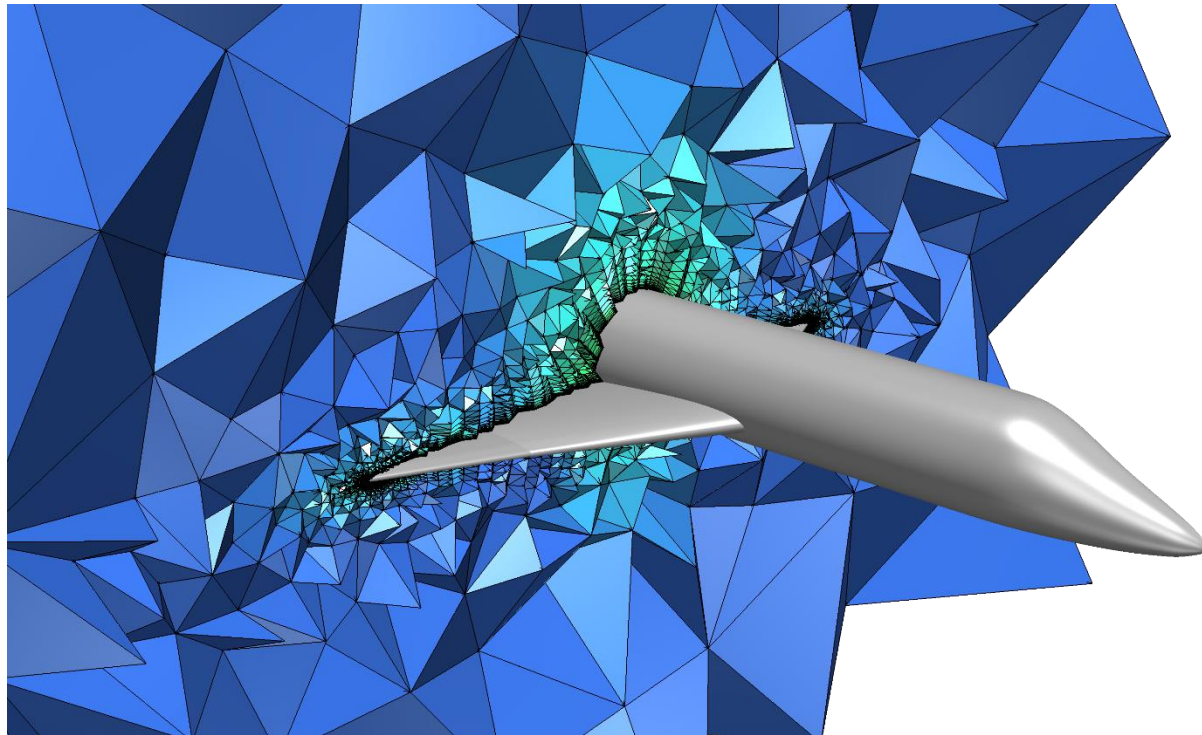
$$\Pi_C(\mathbf{x}) = \frac{1}{2} \left( \Pi_{\mathcal{S}_1}(\Pi_{\mathcal{S}_2}(\mathbf{x})) + \Pi_{\mathcal{S}_2}(\Pi_{\mathcal{S}_1}(\mathbf{x})) \right)$$



# Parallel Distribution of input & output

- **Input data:**
  - Marked linear mesh
  - Virtual CAD model
- **Linear & high-order meshes:**
  - Each processor owns a set of elements and nodes
  - Each processor projects his boundary nodes
- **Virtual model:**
  - Each processor has a copy of the geometry
  - Easy to distribute, just read the CAD file
  - Processors have enough memory for this approach





# Large-scale distributed curving

(Ruiz-Gironés, Roca IMR'19 & CAD'22)

- **Reduce computational time**
- **Reduce memory footprint**
- **Reduce energy consumption**

# Lagrange Multiplier Approximation

- **In penalty method:** for  $\mu$  large enough, Lagrange multiplier is like

$$\lambda \simeq -2\mu (T\phi - g_D(T\phi))$$

- **Function over the mesh boundary:**
  - When converging, doubling  $\mu$  halves the constraint values
- **We use this to improve the solver:**
  - Given a target constraint norm, which  $\mu$  we need?

# p-Continuation: Less DOF's & Sparser Matrices

**Idea:** Use a lower degree solution as an initial approximation

## Implementation:

- Increase polynomial degree when boundary constraint is “*good enough*”

$$\alpha \varepsilon^p < \varepsilon^{p+1}, \quad \varepsilon^p = \|\mathbf{T}\phi^p - \mathbf{g}_D(\mathbf{T}\phi^p)\|_{\partial\mathcal{M}_I}$$

- Approximate next penalty parameter using constraint norm

$$\mu^{p+1} = \mu^p \frac{\|\mathbf{T}\phi^p - \mathbf{g}_D(\mathbf{T}\phi^p)\|_{\partial\mathcal{M}_I}}{\|\mathbf{T}\phi^{p+1} - \mathbf{g}_D(\mathbf{T}\phi^{p+1})\|_{\partial\mathcal{M}_I}}$$

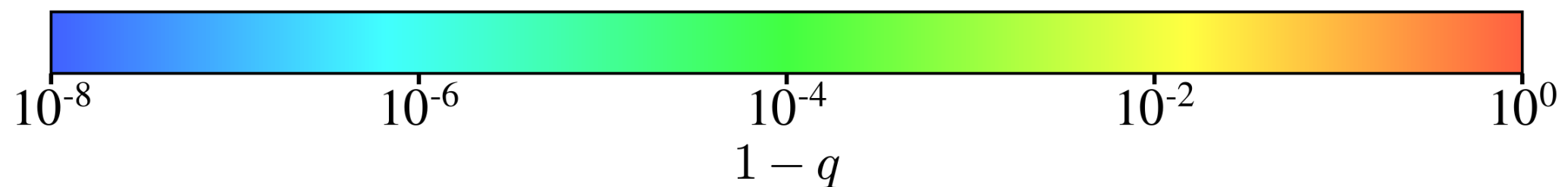
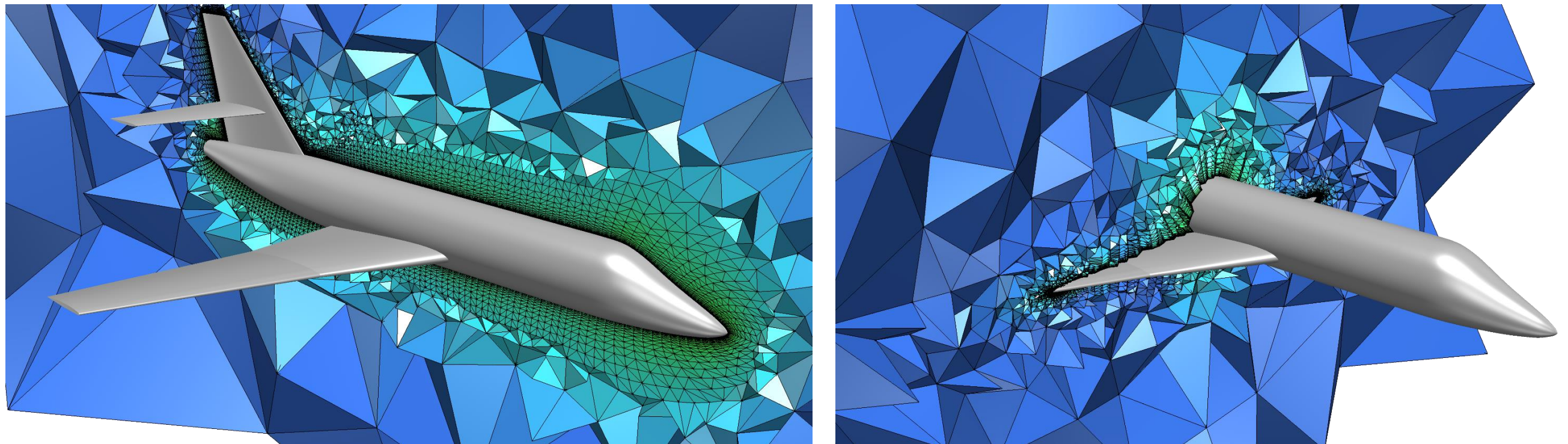


# Examples

**p-continuation:** Falcon aircraft

**Mesh:** Degree 4, 4M elements, boundary layer stretching 1:400

**Optimization:** 2400 cores, RASDD(1) - SSOR(2)



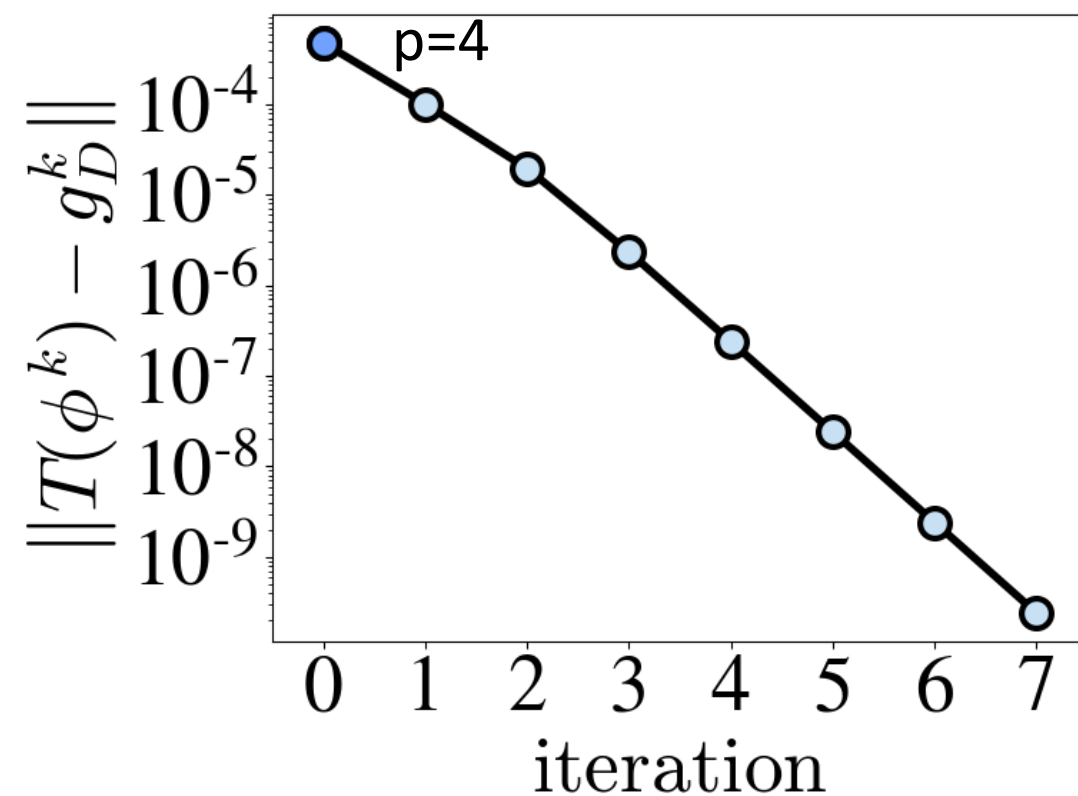
# Examples

**p-continuation:** Falcon aircraft

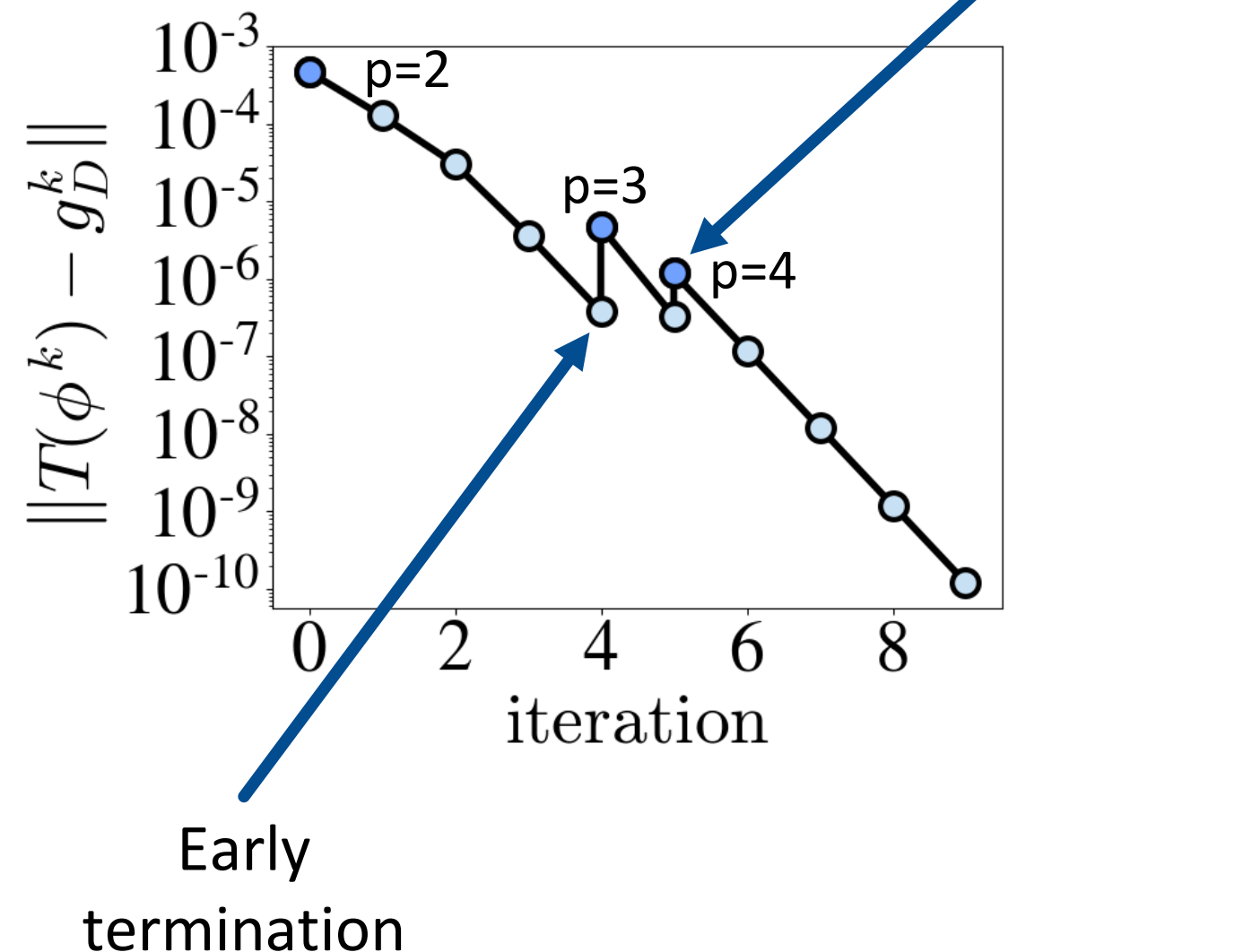
**Mesh:** Degree 4, 4M elements, boundary layer stretching 1:400

**Optimization:** 2400 cores, RASDD(1) - SSOR(2)

No p-continuation



p-continuation



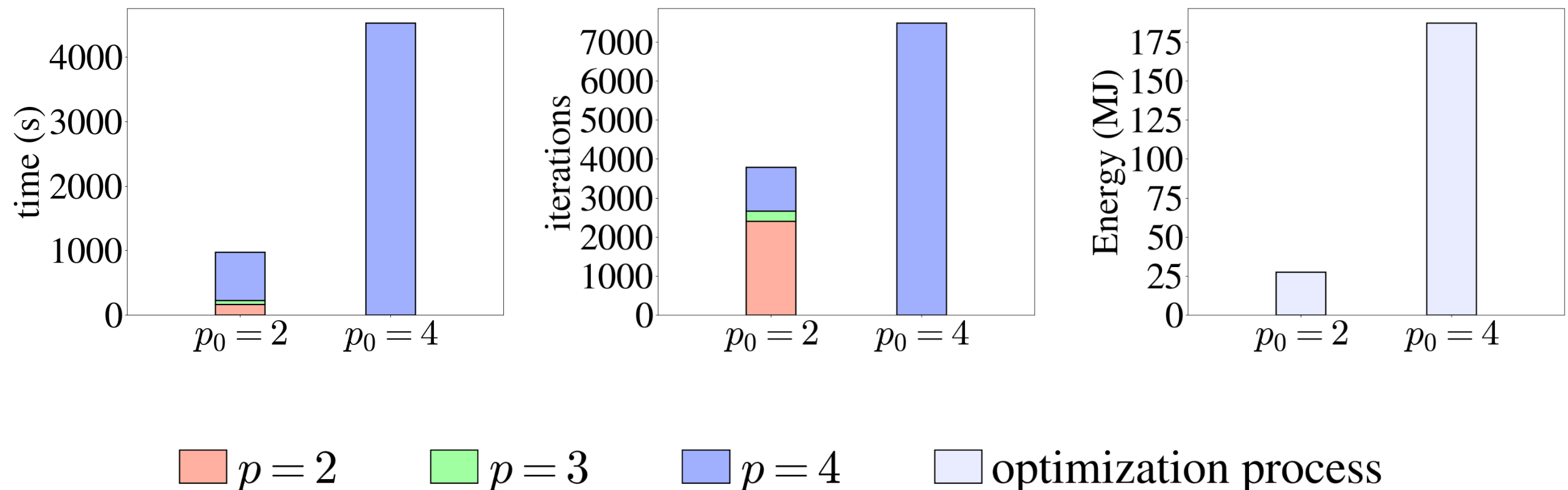


# Examples

**p-continuation:** Falcon aircraft

**Mesh:** Degree 4, 4M elements, boundary layer stretching 1:400

**Optimization:** 2400 cores, RASDD(1) - SSOR(2)



p-continuation technique allows:

- 4 times reduction in time
- 8 times reduction in energy

# Penalty Estimation: Less non-linear problems

Optimal penalty parameter

$$\mu^* = \mu_k m^* = \mu_k 1.01 \frac{\varepsilon^*}{\varepsilon_k}$$

Convergence indicator

$$s_k = \frac{\mu_{k-1}}{\mu_k} \frac{\varepsilon_{k-1}}{\varepsilon_k} \quad e_k = \max\{s_k, 1/s_k\} - 1$$

Next penalty parameter

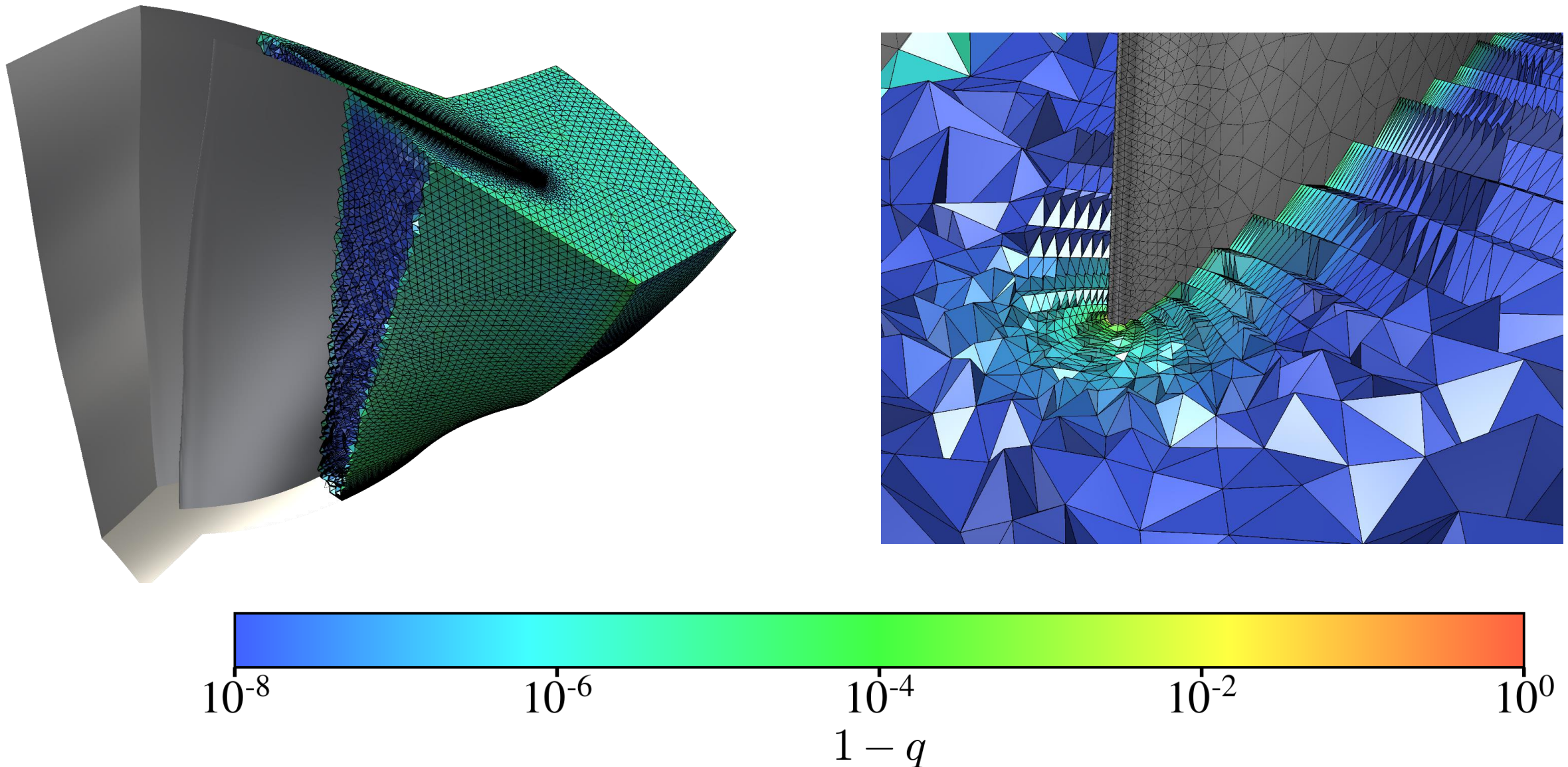
$$m_k = \max\{10, 1/e_k\}$$

$$\mu_{k+1} = \mu_k \min\{m^*, m_k\}$$

# Example: Periodic Mesh for the Rotor 67

Set periodic condition in target boundary

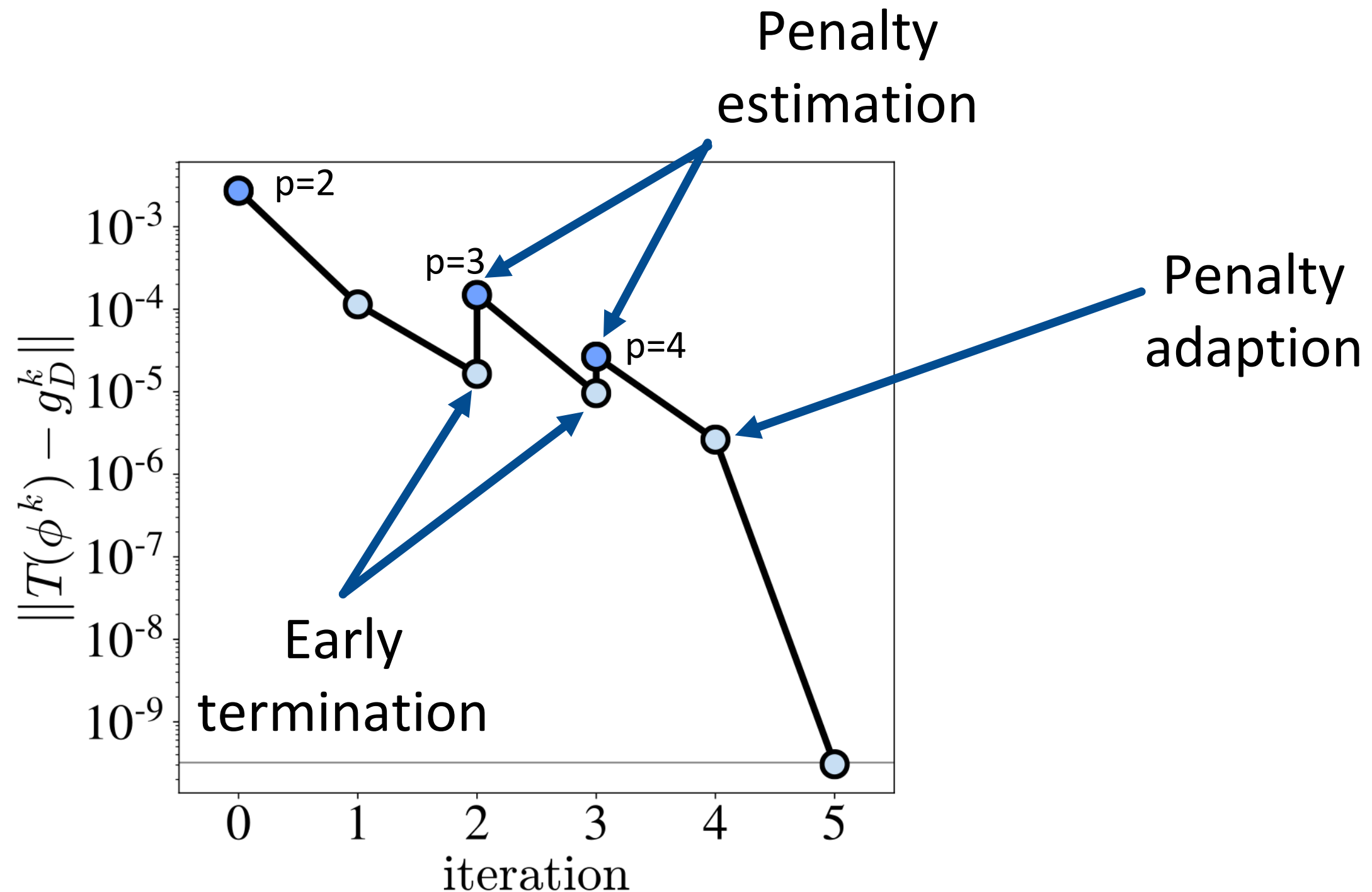
- **Mesh:**  $p=4$ , 3.6M elements, boundary layer stretching 1:25
- **Optimization:** 768 cores, 52 minutes, 31.5MJ
- **Metrics:**  $\min Q = 0.987$ ,  $\text{avg\_dist} = 1.5 \cdot 10^{-8}$ ,  $\text{max\_dist} = 2.91 \cdot 10^{-5}$



# Example: Periodic Mesh for the Rotor 67

**Periodic mesh:** Impose periodic boundary condition

**Mesh:**  $p=4$ , 3.6M elements, boundary layer stretching 1:25



# Forcing Term: Less Linear Iterations

**Adapt linear solver tolerance:** From loose to tight tolerance

Track the progress of the optimization: use boundary constraint

$$t_k = \frac{\log\left(\frac{\varepsilon_0}{\varepsilon_k/m_k}\right)}{\log\left(\frac{\varepsilon_0}{\varepsilon^*}\right)}$$

Linear solver tolerance

$$\delta = \delta_{\text{loose}}^{1-t_k} \cdot \delta_{\text{tight}}^{t_k}$$



# Pre-conditioned GMRES: 3 Times Less Memory

- Separate the linear problem in smaller blocks: x, y, z coordinates

$$\mathbf{H} = \begin{pmatrix} \mathbf{H}_{1,1} & \mathbf{H}_{1,2} & \mathbf{H}_{1,3} \\ \mathbf{H}_{2,1} & \mathbf{H}_{2,2} & \mathbf{H}_{2,3} \\ \mathbf{H}_{3,1} & \mathbf{H}_{3,2} & \mathbf{H}_{3,3} \end{pmatrix} \quad \mathbf{b} = \begin{pmatrix} \mathbf{b}_1 \\ \mathbf{b}_2 \\ \mathbf{b}_3 \end{pmatrix}$$

- Block-based SOR and forward substitution

$$\begin{pmatrix} \mathbf{H}_{1,1} & \mathbf{0} & \mathbf{0} \\ \mathbf{H}_{2,1} & \mathbf{H}_{2,2} & \mathbf{0} \\ \mathbf{H}_{3,1} & \mathbf{H}_{3,2} & \mathbf{H}_{3,3} \end{pmatrix} \begin{pmatrix} \mathbf{x}_1^{l+1} \\ \mathbf{x}_2^{l+1} \\ \mathbf{x}_3^{l+1} \end{pmatrix} = \begin{pmatrix} \mathbf{b}_1 \\ \mathbf{b}_2 \\ \mathbf{b}_3 \end{pmatrix} - \begin{pmatrix} \mathbf{0} & \mathbf{H}_{1,2} & \mathbf{H}_{1,3} \\ \mathbf{0} & \mathbf{0} & \mathbf{H}_{2,3} \\ \mathbf{0} & \mathbf{0} & \mathbf{0} \end{pmatrix} \begin{pmatrix} \mathbf{x}_1^l \\ \mathbf{x}_2^l \\ \mathbf{x}_3^l \end{pmatrix}$$

- Each block: GMRES preconditioned with RASDD(1)+SSOR(2)
- Store the 3 diagonal blocks and use matrix-free products for the rest
- 3 iteration of block-SOR,  $\delta_{pre} = \delta^{1/2}$



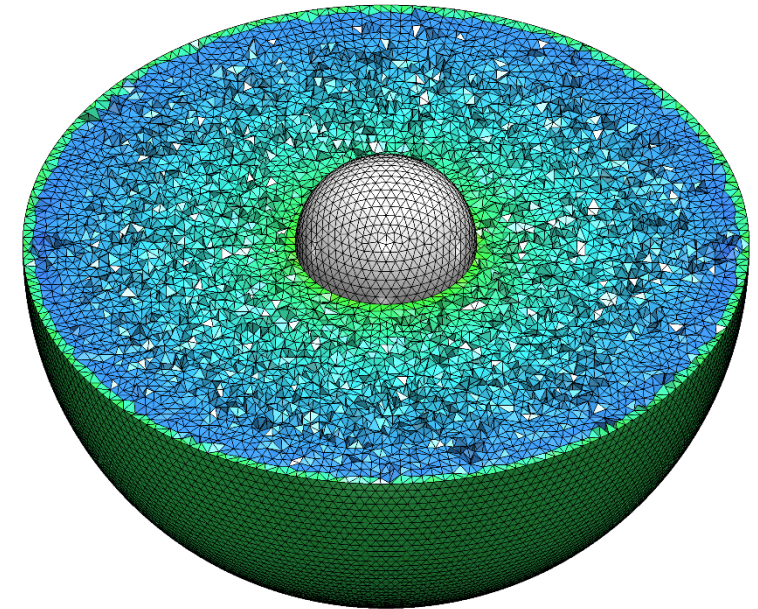
# Example: Influence of the improvements

## Uniform Mesh for a sphere

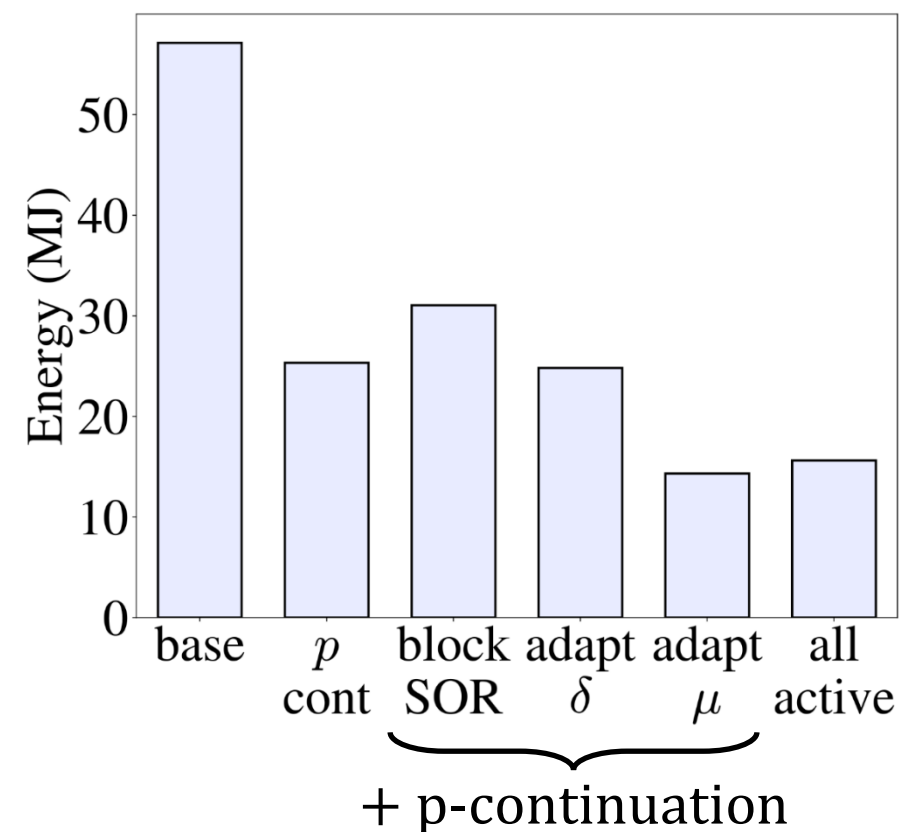
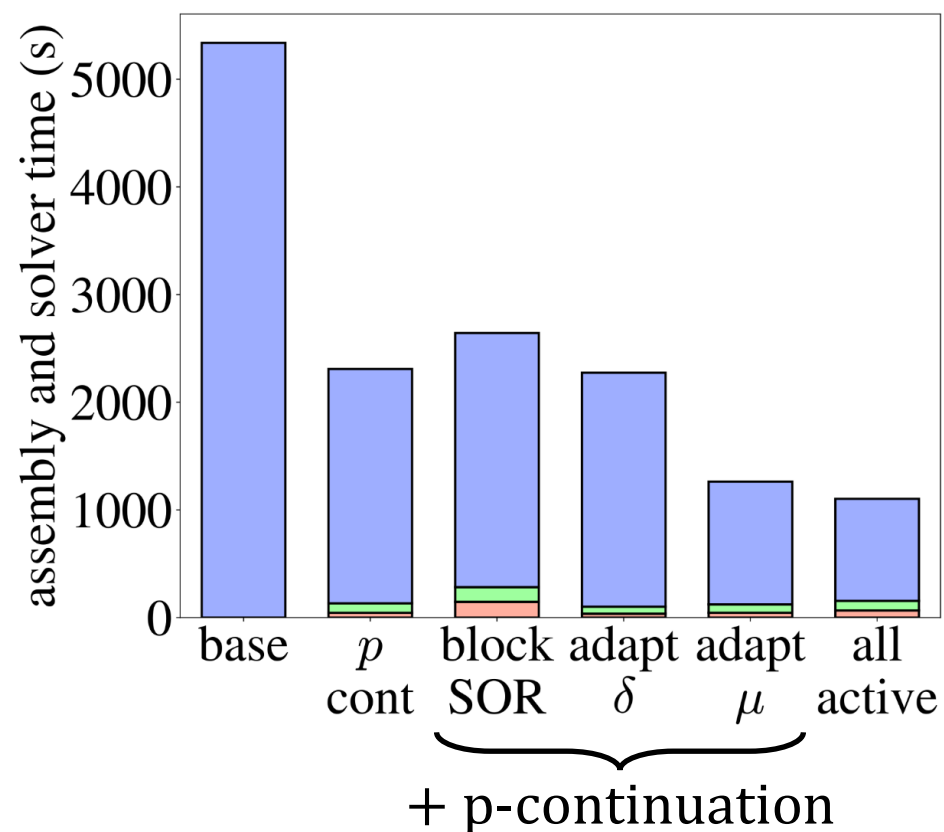
Isotropic mesh

1.44M elements,  $p=4$

768 processors



■  $p = 2$     ■  $p = 3$     ■  $p = 4$     ■ optimization process



# Distributed Solver: Concluding Remarks

- **Key Improvements:**
  - p-continuation
  - Penalty parameter adaption
  - Block-SOR pre-conditioner
  - Forcing term (only for  $p=2$ )
- **Improvements:**
  - Decrease time and energy: 4 times
  - Decrease memory footprint: 3 times

**3 times larger meshes with  
the same resources**



# High-Lift Prediction Workshop

(Ruiz-Gironés, Roca AIAA'22)

- **Pre-process**
- **Post-process**
- **Software, libraries, and languages**

# HLPW: Pre-processing for curving-friendly inputs

- **Setup the simulation intent:** repair geometry & virtual model
- **Linear mesh generation:**
  - **Element size:** Simulation and geometry accuracy
  - **Curving:** curving-friendly mesh leads to easy curving process
- **Convert sequential inputs to parallel inputs**
  - Sequential inputs are bottlenecks
  - Create a hdf5 parallel input

# HLPW: Check curved mesh and create output file

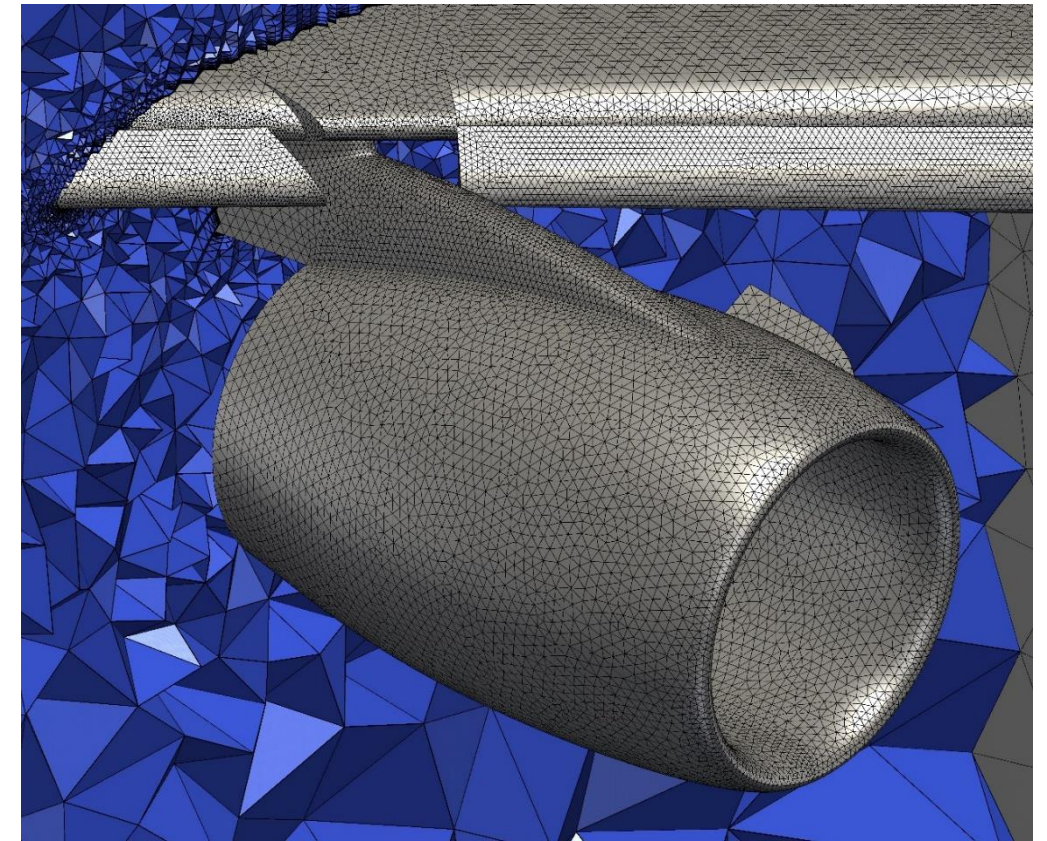
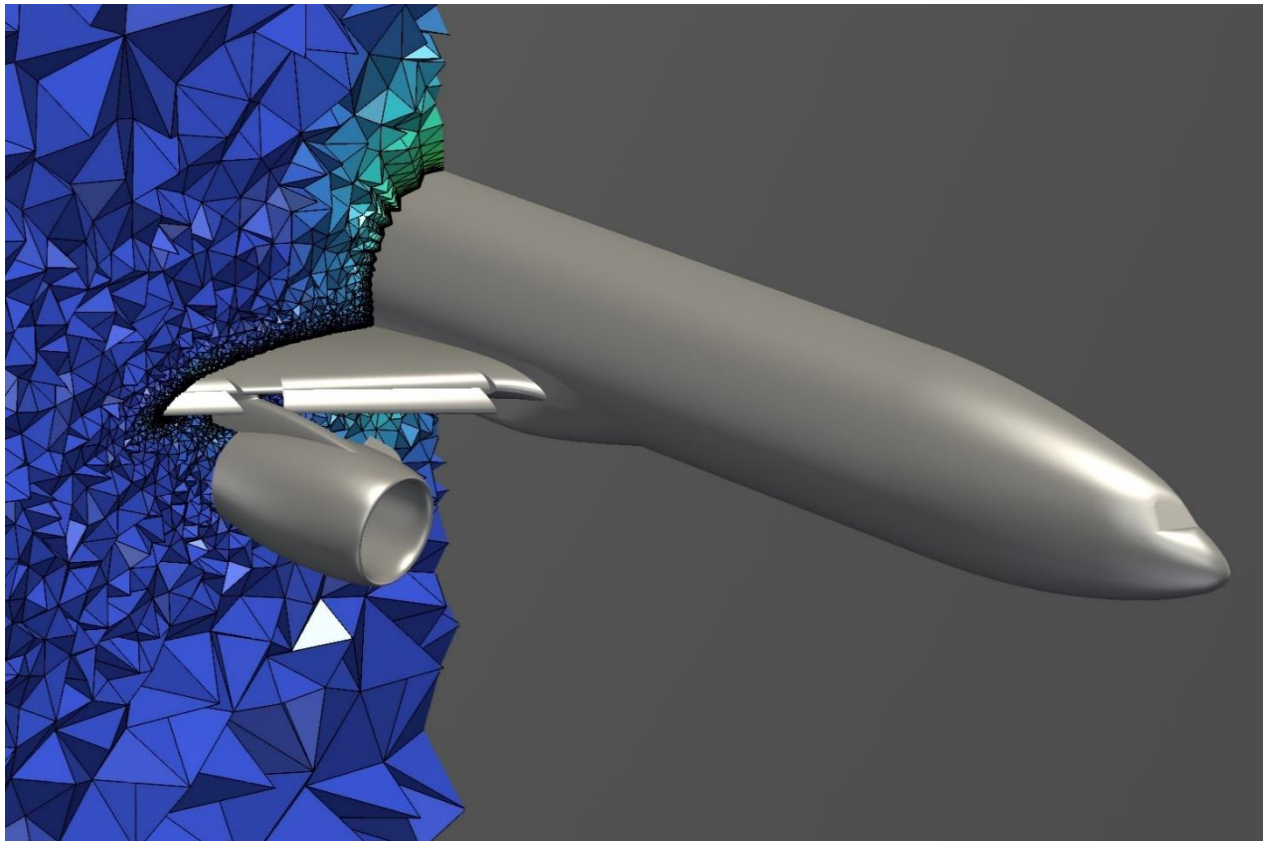
- **Mesh validity and quality:** Numeric validation
- **Visual inspection:** Paraview in distributed parallel  
Locate low-quality and low-accuracy elements
- **Curving iterative process:**  
Remesh low-quality and low-accuracy elements
- **Create output file:** python wrapper of cgns library

- **Virtual model & linear mesh:** Pointwise
- **Distributed solver:** our python implementation with FEniCS library
- **CAD engine:** our python wrapper of Project Geode / OpenCASCADE
- **Linear solver library:** petsc4py interface to PETSc
- **Distributed parallel solver:** running on MareNostrum 4
- **Visualization:** distributed parallel Paraview
- **cgns output:** our python wrapper of cgns library



# Example: CRM-HL of the 4<sup>th</sup> HLPW

- **Mesh:**  $p=2$  &  $p=3$ , 8M elements, boundary layer stretching 1:250
- **Accuracy:** relative to aircraft length  $\sim 10^{-7} - 10^{-6}$
- **Computational resources:** 768 processors
  - $p = 2 \rightarrow 12$  minutes
  - $p = 3 \rightarrow 48$  minutes



Our meshes provided the best match with experimental results  
(ZJ Wang AIAA'22)

# Our Participation in HLPW: Concluding Remarks

- **Preparing curving-friendly inputs takes days** (human labor)
  - Tune the virtual model & linear mesh → Iterative process
  - Curving-friendly inputs → High-quality mesh in a short time
- **Mesh curving for the CRM-HL takes minutes** (computing wall time)
  - We generate larger meshes than the CFD community wants to run
  - Curving is a steady-state problem with less unknowns than CFD
- **You can try our meshes!**
  - Free to download in the 4<sup>th</sup> & 5<sup>th</sup> HLPW websites

**Our meshes provided the best match with experimental results**



# Summary & Conclusions

# Summary & conclusions: Large-Scale Curving

- **Mesh curving constrained formulation:**
  - Always numerically valid
  - Optimal quality
  - Approximates target geometry
  - Tightly converged
- **Complex geometry in parallel:** mesh approximates virtual geometry
- **Large-scale curving:** 3 times larger meshes on thousands of cores
- **High-lift prediction:** Our meshes lead to best match with experiments

**Our curving enables high-fidelity simulations on complex geometries**



# Thank you for your attention!



European Research Council  
Established by the European Commission

This project has received funding from the European Research Council (ERC) under the European Union's Horizon 2020 research and innovation programme under grant agreement No 715546.